

Amendments To The Claims:

Please amend the claims as shown. Applicant reserves the right to pursue any canceled claims at a later date.

1. (currently amended) An annular combustion chamber for a gas turbine extending in the axial direction, comprising; ~~whereby the annular combustion chamber extends in an axial direction, encloses a combustor, and has on its inside facing the combustor a bearing structure on which a lining element secured to this lines the annular combustion chamber,~~

~~— wherein~~

~~— on a rear side facing away from the combustor of two edge areas extending in the axial direction on the lining element a plurality of interlocking devices are located which have a width, and in that the lining element is secured to the bearing structure such that in order to release the lining element from the bearing structure the lining element (10) is moved by the extent of the width of the interlocking means in the axial direction~~

a combustor enclosed by the combustion chamber;

a bearing structure arranged on the inside of the combustion chamber and facing the combustor;

a lining element having a rear side facing away from the combustor, two edge areas extending in the axial direction, and secured to the bearing structure that lines the annular combustion chamber; and

a plurality of interlocking devices having a width arranged on the lining element such that in order to release the lining element from the bearing structure the lining element is moved by the extent of the width of the interlocking means in the axial direction, wherein

two interlocking devices of the lining element that are immediately adjacent in the axial direction have an axial spacing which is identical to or greater than the width of the interlocking devices.

2. (currently amended) ~~An~~ The annular combustion chamber according to Claim 1, wherein a further plurality of interlocking devices are located as a central support midway between two edge areas of the lining element running in the axial direction.

3. (canceled)

4. (currently amended) ~~An~~ The annular combustion chamber according to ~~one of the~~ Claims 1, wherein each interlocking device has the identical hook width.

5. (currently amended) ~~An~~ The annular combustion chamber according to Claim 1, wherein two interlocking ~~means~~ devices of the lining element that are immediately adjacent in the axial direction have a an axial spacing which is twice the width of an interlocking device.

6. (currently amended) ~~An~~ The annular combustion chamber according to Claim 1, wherein two interlocking devices of the lining element that are immediately adjacent in the axial direction have a an axial spacing which is three times the width of an interlocking device.

7. (currently amended) ~~An~~ The annular combustion chamber according to Claim 5, wherein each spacing between two interlocking devices of the lining element that are immediately adjacent in the axial direction is identical.

8. (currently amended) ~~An~~ The annular combustion chamber according to Claim 1, wherein the lining element has stiffening ribs running in the circumferential direction of the annular combustion chamber on its rear side facing away from the combustor.

9. (currently amended) ~~An~~ The annular combustion chamber according to Claim 8, wherein the stiffening rib is distanced from the interlocking means.

10. (currently amended) ~~An~~ The annular combustion chamber according to Claim 1, wherein the interlocking ~~means~~ devices are L- and/or T-shaped.

11. (currently amended) A gas turbine with an annular combustion chamber wherein the combustion chamber comprises:

~~a combustor enclosed within the chamber;~~

~~a bearing structure arranged on an inside facing of the chamber;~~

~~a lining element secured to the bearing structure and lining the chamber; and~~
~~a plurality of interlocking devices located on a rear side of two edge areas extending in an axial direction on the lining element and having a width,~~
~~wherein the lining element is secured to the bearing structure such that in order to release the lining element from the bearing structure the lining element is moved the width of the interlocking device in the axial direction.~~

a combustor enclosed by the combustion chamber;
a bearing structure arranged on the inside of the combustion chamber and facing the combustor;
a lining element having a rear side facing away from the combustor, two edge areas extending in the axial direction, and secured to the bearing structure that lines the annular combustion chamber; and
a plurality of interlocking devices having a width arranged on the lining element such that in order to release the lining element from the bearing structure the lining element is moved by the extent of the width of the interlocking means in the axial direction, wherein
two interlocking devices of the lining element that are immediately adjacent in the axial direction have an axial spacing which is identical to or greater than the width of the interlocking devices.

12. (currently amended) A combustion chamber for a gas turbine, comprising:

~~a combustor enclosed within the chamber;~~
~~a bearing structure arranged on an inside facing of the chamber;~~
~~a lining element secured to the bearing structure and lining the chamber; and~~
~~a plurality of interlocking devices located on a rear side of an edge area extending in an axial direction on the lining element and having a width,~~
~~wherein the lining element is secured to the bearing structure such that in order to release the lining element from the bearing structure the lining element is moved the width of the interlocking device in the axial direction.~~

a combustor enclosed by the combustion chamber;
a bearing structure arranged on the inside of the combustion chamber and facing the combustor;

a lining element having a rear side facing away from the combustor, two edge areas extending in the axial direction, and secured to the bearing structure that lines the annular combustion chamber; and

a plurality of interlocking devices having a width arranged on the lining element such that in order to release the lining element from the bearing structure the lining element is moved by the extent of the width of the interlocking means in the axial direction, wherein

two interlocking devices of the lining element that are immediately adjacent in the axial direction have an axial spacing which is identical to or greater than the width of the interlocking devices.

13. (currently amended) ~~An~~ The annular combustion chamber according to Claim 12, wherein the interlocking devices face away from the combustor.

14. (currently amended) ~~An~~ The annular combustion chamber according to Claim 12, wherein the interlocking device is a hook.

15. (currently amended) ~~An~~ The annular combustion chamber according to Claim 12, wherein the width is the hook width.

16. (currently amended) ~~An~~ The annular combustion chamber according to Claim 1, wherein the interlocking device is a hook.

17. (currently amended) ~~An~~ The annular combustion chamber according to Claim 1, wherein the width is the hook width.

18. (currently amended) ~~An~~ The annular combustion chamber according to Claim 2, wherein two interlocking ~~means~~ devices of the lining element that are immediately adjacent in the axial direction have a an axial spacing which is identical to or greater than the hook width of the interlocking means.

19. (currently amended) ~~An~~ The annular combustion chamber according to ~~one of the~~ Claims 2, wherein each interlocking ~~means~~ device has the identical hook width.

20. (currently amended) ~~An~~ The annular combustion chamber according to Claim 2, wherein two interlocking ~~means~~ devices of the lining element that are immediately adjacent in the axial direction have a an axial spacing which is twice the hook width of an interlocking device.